



ERRATUM

T. Q. WANG AND S. ZHOU *Journal of Sound and Vibration* 1998 **209**(2), 299–316, Investigation on sound field model of propeller aircraft—the effect of vibrating fuselage boundary (sv971232) and **209**(2), 317–328, Investigation on sound field model of propeller aircraft—the effect of rigid fuselage boundary (sv971231).

The two companion papers were not published in the correct order; paper sv971231 should have preceded paper sv971232. The issue number in reference [1] of paper sv971232 should have read **209**(2), referring the reader to the companion paper published in the same issue. In the authors' address on both papers the full country name should have read Peoples' Republic of China. The publisher apologizes for these errors.

G. CHAKRABORTY, A. K. MALLIK AND H. HATWAL *Journal of Sound and Vibration* 1998 **210**(1), 19–36, Normal modes and near-resonance response of beams with non-linear effects (sv971284)

1. The second sentence of section 2.2, “The linear normal modes... $\varepsilon = 0$.” is to be replaced by “It is to be noted that for the linear case, the longitudinal and transverse motions are uncoupled.”
2. In the caption for Figure 1(a), “ Δ , linear theory,” is to be added along with the other legends.
3. Equations (B4) and (B5) in Appendix B are to be replaced, respectively, by

$$\psi_n = \sin n\pi x + \varepsilon^2 \frac{3}{160} a^2 (n\pi)^2 \sin 3n\pi x, \quad (\text{B4})$$

$$\omega_n = (n\pi)^4 \left[1 + \frac{3}{8} \varepsilon a^2 - \frac{1}{2} \varepsilon^2 a^2 (n\pi)^2 \right]. \quad (\text{B5})$$